

On the Nature of Long-Period X-ray Pulsars

Nazar R. Ikhsanov

Pulkovo Observatory, St. Petersburg, Russia

(Marie Curie Fellow)

- Long-period X-ray Pulsars (LPXPs)
- Evolutionary tracks of massive X-ray binaries
- Conclusions



High Mass X-ray Binaries

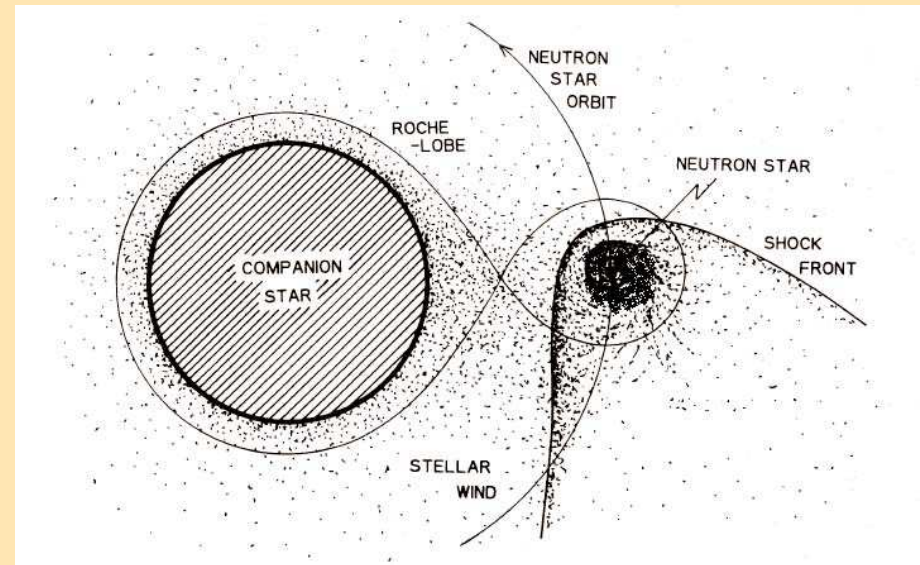
★ Wind-fed Accretion

$$R_{\text{opt}} < R_{\text{Roche}}$$

$$\dot{m}_c = \pi r_\alpha^2 \rho_\infty V_{\text{rel}}$$

$$r_\alpha = \frac{2GM_{\text{ns}}}{V_{\text{rel}}^2}$$

$$|\vec{V}_{\text{rel}}|^2 = |\vec{V}_{\text{ns}} + \vec{V}_w|^2 + V_s^2$$

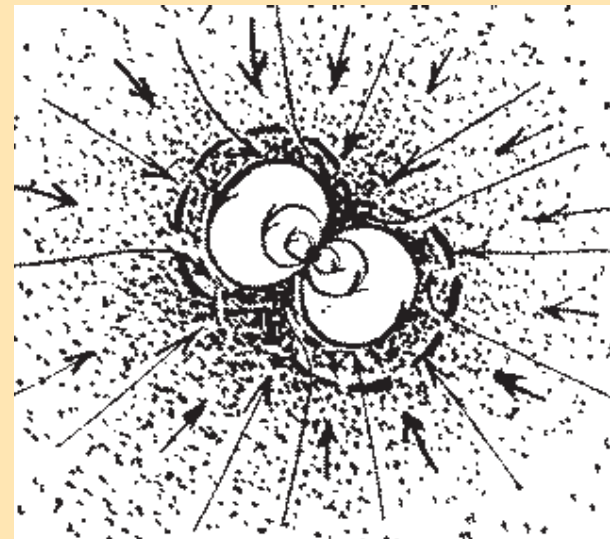


★ Accretion-powered pulsar

$$\dot{m}_a = \frac{L_x R_{\text{ns}}}{GM_{\text{ns}}}$$

$$B(R_{\text{ns}}) \sim 10^{12} \text{ G}$$

(Cyclotron line in X-ray spectrum)



Long-period X-ray pulsars

Name	Sp. type	P_s , s	P_{orb} , d	$\log L_x$
J170006-4157	—	715	—	34.7
0352+309 (X Per)	B0 Ve (O9.5 IIIe)	837	250	34.7 – 35.5
J1037.5-5647	B0 V-IIIe	862	—	34-35
J2239.3+6116	B0 Ve	1247	262.2	33 – 36
J0103.6-7201	O5 Ve	1323	—	35.3 – 36.8
J0146.9+6121 (V831 Cas)	B1 III-Ve	1412	—	34.6 – 36
2S 0114+650	B1 Ia	10008	11.6	34.7 – 35

Canonical Evolutionary Track

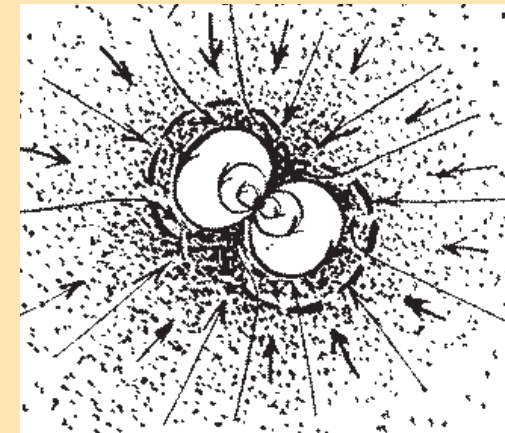
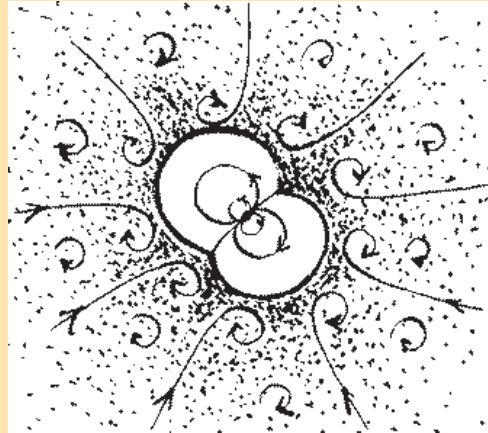
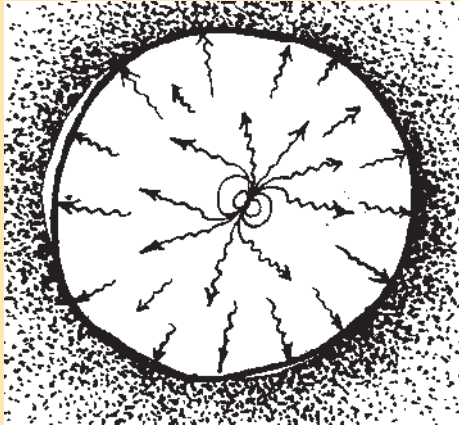
Ejector



Propeller



Accretor



Assumption: Transition to Accretor occurs at $R_m = R_{\text{cor}}$

$$P_{\text{max}} \lesssim \underline{15 \text{ s}} \times m^{-2/7} R_6^{-3/7} L_{35}^{-3/7} B_{12}^{6/7}$$

$$L_x < \underline{10^{31} \text{ erg/s}} \times m^{-2/3} P_{700}^{-7/3} B_{12}^2$$

Something is incorrect...

2S 0114+650 / LSI +65010

Neutron star

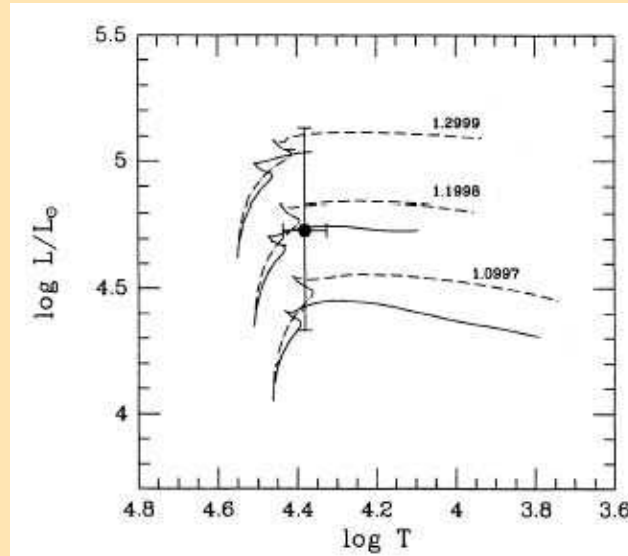
Evolutionary track

Normal companion

$$P_s \simeq 10008 \text{ s}$$

$$\dot{P} \simeq -10^{-7} \text{ s/s}$$

$$\frac{P_s}{\dot{P}} \simeq 3000 \text{ yr}$$



Observed: B1 Ia

Progenitor: O9.5 V

$$\frac{\dot{M}_{\text{loss}}(\text{B1 Ia})}{\dot{M}_{\text{loss}}(\text{O9.5 V})} \sim 10$$

★ Assumption about the spin-down of the NS in the accretor state is not effective

★ Magnetar: $B_0 \sim 10^{15} \text{ G} \rightarrow B(t = 10^5 \text{ yr}) \sim 10^{14} \text{ G}$. Putting $P_{\text{max}} = P_{\text{cd}}(R_m = R_{\text{cor}})$, one finds:

$$\dot{m}_c(t_{\text{sd}}) \lesssim 2 \times 10^{13} \text{ g/s} \times m^{-5/3} \mu_{32}^2 \left(\frac{P_{\text{cd}}}{10^4 \text{ s}} \right)^{-7/3} \Rightarrow \frac{\dot{m}_a}{\dot{m}_c(t_{\text{sd}})} \simeq 250$$

Accretion flow geometry

★ Evolution of the spin period of accretion-powered pulsars:

$$2\pi I \frac{d}{dt} \left(\frac{1}{P_s} \right) = K_{\text{su}} + K_{\text{sd}} \quad \left| \quad \begin{array}{l} K_{\text{sd}} = -k_t (\mu^2 / R_c^3) \\ K_{\text{su}} = \begin{cases} \dot{m}_a \sqrt{GM_{\text{ns}} R_m}, & \text{disk accretion,} \\ \frac{1}{4} \xi \dot{m}_a \Omega_{\text{orb}} r_a^2, & \text{spherical accretion.} \end{cases} \end{array} \right.$$

★ Equilibrium period: $P_{\text{eq}} = \begin{cases} 10 \text{ s} \times m^{-5/7} \mu_{30}^{6/7} \dot{m}_{15}^{-3/7}, & \text{disk accretion,} \\ 900 \text{ s} \times m^{-3/2} \mu_{30} \dot{m}_{15}^{-1/2} V_{500}^2 P_{250}^{1/2} \xi_{0.2}^{-1/2}, & \text{spherical accretion} \end{cases}$

★ For the disk accretion case: $P_{\text{eq}} \ll P_s \Rightarrow \begin{cases} \dot{P} \simeq P_s^2 \frac{\dot{m}_a \sqrt{GM_{\text{ns}} r_m}}{2\pi I} \sim -5 \times 10^{-8} \text{ s/s,} \\ \Delta t \simeq \frac{2\pi I}{\dot{m}_a \sqrt{GM_{\text{ns}} r_m}} \left(\frac{1}{P_s} - \frac{1}{P_0} \right) \lesssim 100 - 1000 \text{ yr} \end{cases}$

I. Regular spin-up at a high rate

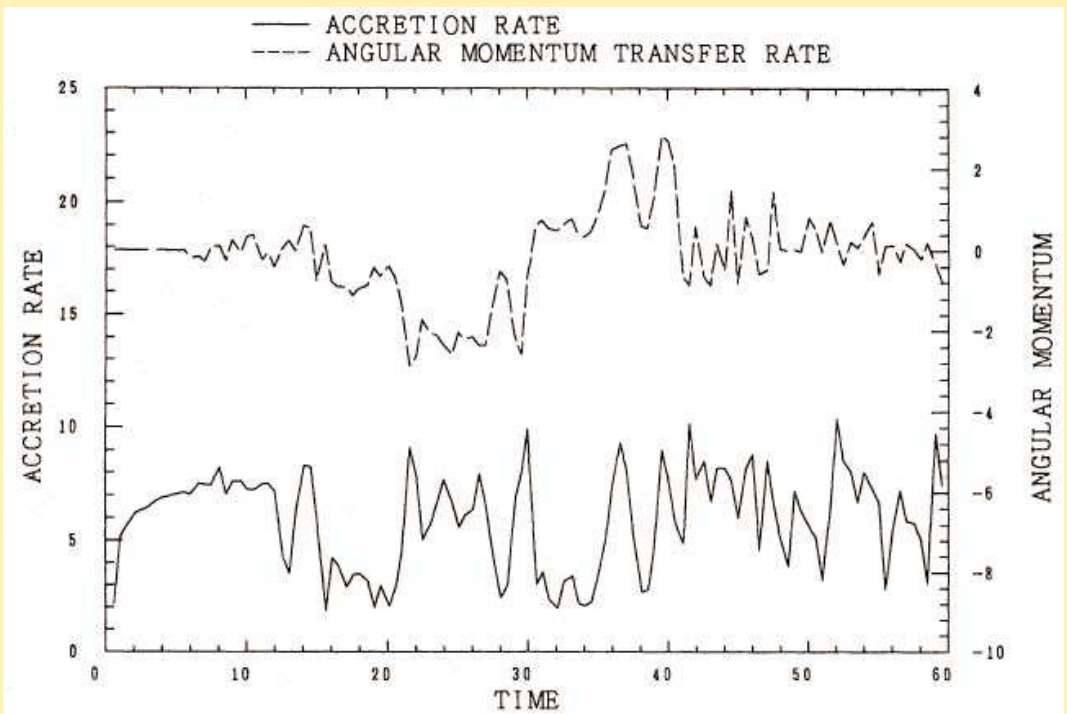
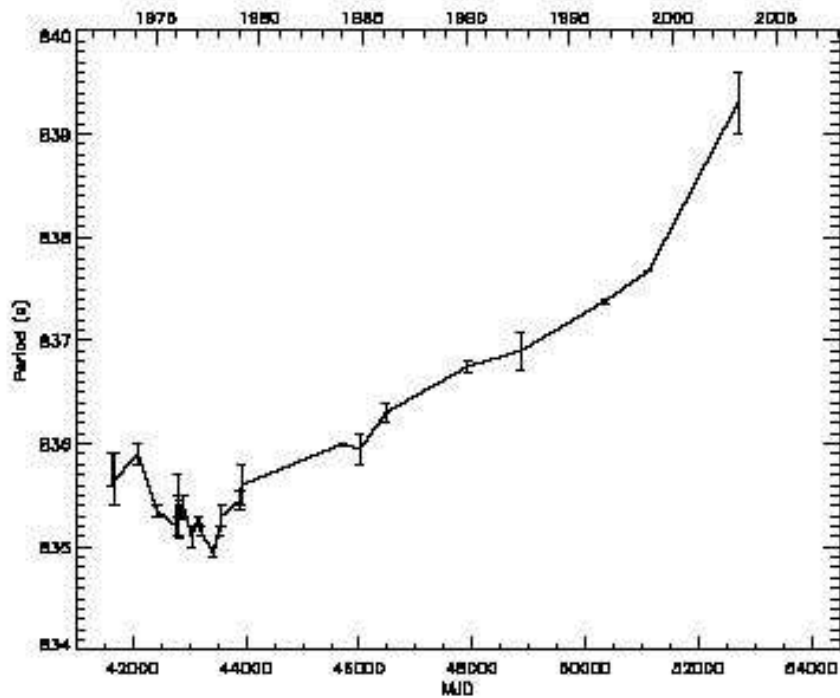
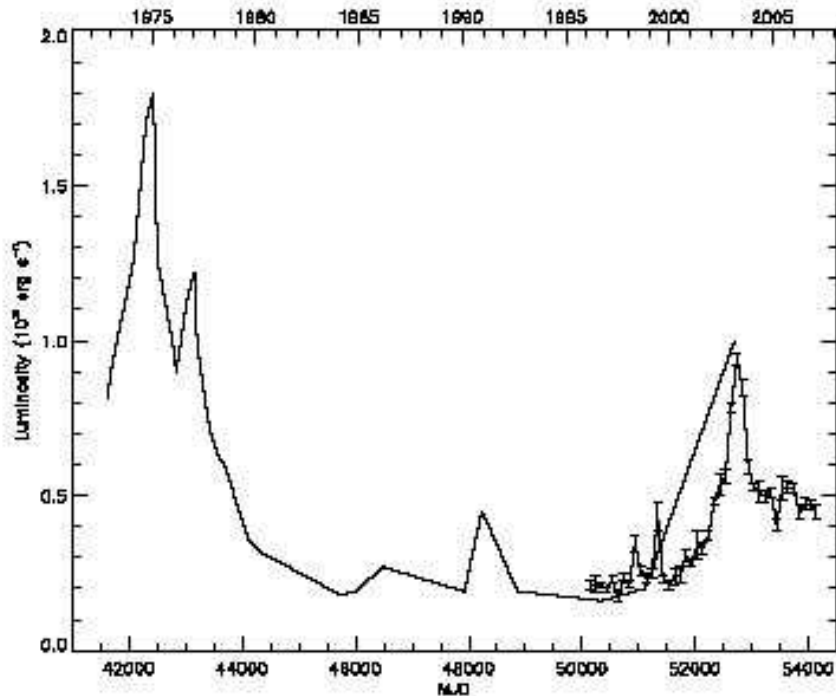
II. Very short life-time of LPXPs

X Persei

$$V_{\text{rel}} \simeq 350 - 400 \text{ km/c} \quad \mu \simeq 1.7 \times 10^{30} \text{ } \Gamma_{\text{c}} \cdot \text{cm}^3$$

$$P_{\text{eq}} = 900 \text{ s} \times m^{-3/2} \mu_{30} \dot{m}_{15}^{-1/2} V_{500}^2 P_{250}^{1/2} \xi_{0.2}^{-1/2}$$

Spherical (Bondi) accretion !

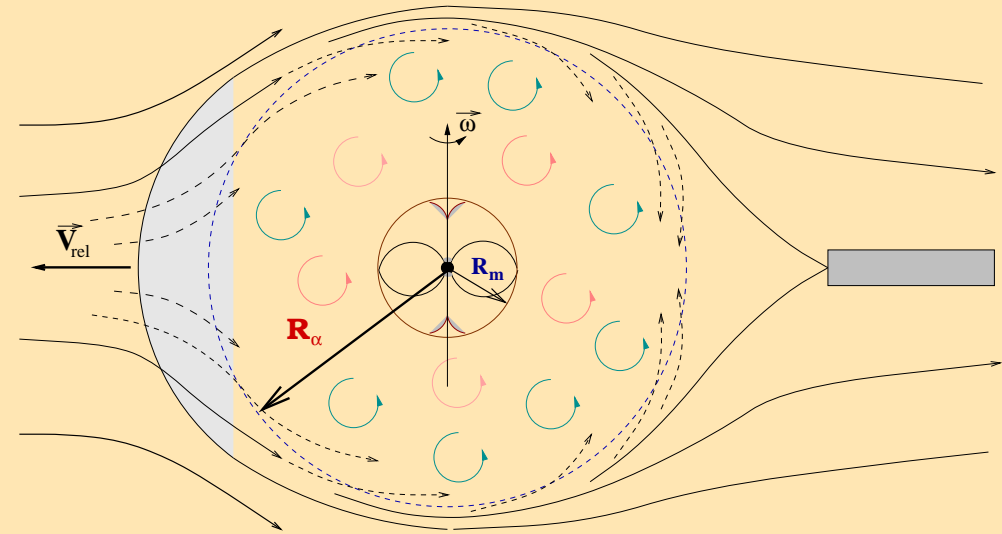


Supersonic propeller $r_m > r_c$ (Spherical accretion flow geometry)

$$R_m \simeq \underline{1.4 \times 10^9 \text{ cm}} \times \mu_{30}^{4/7} \dot{m}_{14.3}^{-2/7} M_{1.4}^{-1/7}$$

$$T_{\text{ff}}(R_m) \simeq \underline{4 \times 10^8 \text{ K}} \times M_{1.4} \left(\frac{r_m}{1.4 \times 10^9 \text{ cm}} \right)^{-1}$$

$$N(R_m) \simeq \underline{4.4 \times 10^{10} \text{ cm}^{-3}} \times \mu_{30}^2 T_{8.6}^{-1} \left(\frac{r_m}{1.4 \times 10^9 \text{ cm}} \right)^{-6}$$



Heating time: $t_{\text{ff}}(R_m) \simeq \underline{3 \text{ s}} \times M_{1.4}^{-1/2} \left(\frac{R_m}{1.4 \times 10^9 \text{ cm}} \right)^{3/2}$

Cooling time: $t_{\text{br}}(R_m) \sim \underline{10^5 \text{ s}} \times T_{8.6}^{1/2} \left(\frac{N(R_m)}{4.4 \times 10^{10} \text{ cm}^{-3}} \right)^{-1}$

The hot envelope forms if

$$\dot{m}_0 \lesssim 3.5 \times 10^{17} \text{ g/s} \times m V_8$$

Correction of the DFP-scenario by a factor of 175 !

Сверхзвуковой пропеллер: Условие образования оболочки

$$\Gamma = \frac{[\text{Темп нагрева конвективных элементов}]}{[\text{Радиационные потери конвективных элементов}]}$$

Условие охлаждения турбулентной оболочки:

$$\Gamma(r) = M_{\text{Mach}}^2 \left[\frac{V_t t_{\text{br}}}{r} \right] = \frac{V_t^3 t_{\text{br}}}{V_s^2 r} < 1, \quad \Gamma \propto r^{-3/2} \quad (1)$$

t_{br} время охлаждения вследствие тормозного излучения:

$$t_{\text{br}} = 6.3 \times 10^4 \left[\frac{T}{10^9 \text{ K}} \right]^{1/2} \left[\frac{n}{10^{11} \text{ cm}^{-3}} \right]^{-1} \text{ c} \quad (2)$$

n – плотность плазмы оболочки:

$$n(r_\alpha) \simeq \frac{\dot{M}_c}{\pi r_\alpha^2 V_\infty m_p}. \quad (3)$$

Скорость турбулентных движений в оболочке, вызванных эффектом пропеллера:

$$V_s \sim V_t \sim V_{\text{ff}}(r) = \sqrt{2GM_{\text{ns}}/r} \quad (4)$$

Вириальная температура:

$$T(r) \sim T_{\text{ff}}(r) = (GM_{\text{ns}}m_p)/(kr) \quad (5)$$

Радиус гравитационного захвата материи нейтронной звездой (радиус Бонди):

$$r_\alpha = (2GM_{\text{ns}})/V_\infty^2 \quad (6)$$

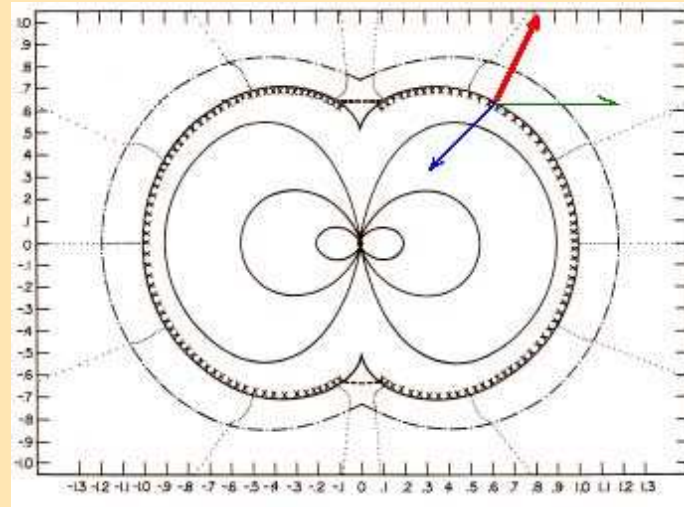
$$\dot{M}_c \lesssim \dot{M}_{\text{max}} = 3.5 \times 10^{17} \left[\frac{M_{\text{ns}}}{M_\odot} \right] \left[\frac{V_\infty}{10^8 \text{ cm s}^{-1}} \right] \Gamma/\text{c}$$

Supersonic propeller \implies Subsonic propeller \implies Accretor

$$a_{\text{curv}} = \frac{V_{T_i}^2(r_m)}{R_{\text{curv}}(\theta)}$$

$$a_{\text{grav}} = \frac{GM_{\text{ns}}}{R_m^2(\theta)}$$

$$a_{\text{cent}} = \Omega_s^2 r_m$$



$$g_{\text{eff}} = \frac{GM_{\text{ns}}}{R_m^2(\theta)} - \frac{V_{T_i}^2(r_m)}{R_{\text{curv}}(\theta)} > 0$$

$$\underline{\underline{T(r_m) \lesssim T_{\text{cr}} \simeq (0.1 - 0.3) T_s(R_m)}}$$

Cooling at R_m dominates heating if

$$P_s \gtrsim P_{\text{br}}(t_{\text{sd}}) \simeq 2000 \text{ s} \times m^{-4/21} \mathfrak{M}_{15}^{-5/7} \left(\frac{B(t_{\text{sd}})}{0.3 B_{\text{cr}}} \right)^{16/21} \left(\frac{R_{\text{ns}}}{10^6 \text{ cm}} \right)^3$$

Correction of DFP-model by a factor of 8 !

$P_s(R_m=R_{\text{cor}}) \ll P_{\text{br}} \longrightarrow$ New state: Subsonic propeller

Дозвуковой пропеллер: Условие существования оболочки

Условие охлаждения турбулентной оболочки:

$$\Gamma(r) = M_{\text{Mach}}^2 \left[\frac{V_t t_{\text{br}}}{r} \right] = \frac{V_t^3 t_{\text{br}}}{V_s^2 r} < 1, \quad \Gamma \propto r^{1/2} \quad (1)$$

t_{br} время охлаждения вследствие тормозного излучения:

$$t_{\text{br}} = 6.3 \times 10^4 \left[\frac{T}{10^9 \text{ K}} \right]^{1/2} \left[\frac{n}{10^{11} \text{ cm}^{-3}} \right]^{-1} \text{ c} \quad (2)$$

n – плотность плазмы на границе магнитосферы:

$$n(r_m) = \frac{\mu^2}{8\pi k T(r_m) r_m^6} \quad (3)$$

Скорость турбулентных движений в оболочке, вызванных эффектом пропеллера:

$$V_t \sim \Omega_s R_m \quad (4)$$

Вириальная температура:

$$T(r) \sim T_{\text{ff}}(r) = (GM_{\text{ns}} m_p)/(kr) \quad (5)$$

Радиус магнитосферы:

$$r_m \equiv \left(\frac{\mu^2}{\dot{M}_c \sqrt{2 G M_{\text{ns}}}} \right)^{2/7} \quad (6)$$

$$P_{\text{br}} \simeq 450 \mu_{30}^{16/21} \left(\frac{M_{\text{ns}}}{M_{\odot}} \right)^{-4/21} \left(\frac{\dot{M}_c}{10^{15} \text{ g s}^{-1}} \right)^{-5/7} \text{ c}$$

LPXPs Formation Scenario: Transition to Accretor at $P_{\max} \simeq P_{\text{br}}$

Name	$\frac{\langle \mathfrak{M} \rangle}{\langle \mathfrak{M}_x \rangle}$	$\frac{B_0}{B_{\text{cr}}}$	$\frac{R_{\text{ns}}}{10^6 \text{ cm}}$
J170006-4157	1	0.08	1
0352+309/X Per	1	0.3	1
J1037.5-5647	1	0.3	1
J2239.3+6116	1	0.1	1
J0103.6-7201	1	0.9	1.3
J0146.9+6121/V831 Cas	1	0.3	1.3
2S 0114+650	1	1	1.3

CONCLUSIONS

- ★ LPXPs are Neutron Stars undergoing spherical accretion
- ★ Spin-down epoch contains four evolutionary states:
 - Ejector →
 - Supersonic propeller →
 - Subsonic propeller →
 - Accretor
- ★ Assumption about supercritical MF of LPXPs is **not necessary**
- ★ A question about the mechanism of accretion in LPXPs is open